

# Série 3b Solutions

## Exercise 3b.1 - Thermal effects

Consider the squared cross-section axial bar of Figure 3b.1. The thermal expansion coefficients of the material is  $10 \cdot 10^{-6} \, K^{-1}$  and its Young modulus is 40 GPa. The initial temperature is room temperature (25°C). The tensile load is 480 N.

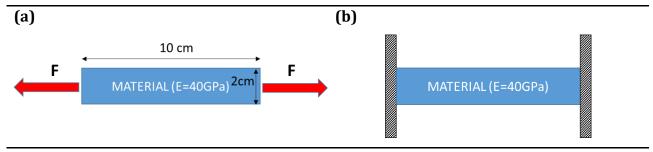


Figure 3b.1 | (a) Loaded bar, (b) Clamped bar.

- We put the bar in a 500°C furnace (Figure 3b.1.a).
  - a) Determine the total longitudinal elongation
  - b) Determine the strain energy density of the bar
- Still in the furnace, we clamp the bar on its longitudinal ends (Figure 3b.1.b) and, from the furnace, put it in liquid Nitrogen (-200°C).
  - NB-<u>It is not anymore submitted to the external load F.</u>
  - c) Determine the value of the induced stress due to temperature in the bar.



### What is given?

Thermal expansion coefficient  $\alpha = 10 \cdot 10^{-6} \, K^{-1}$ 

Young Modulus E=40 GPa

Room temperature  $T_0 = 25 \,^{\circ}C$ 

Furnace temperature  $T_1 = 500 \, ^{\circ}C$ 

Liquid nitrogen environment temperature: Room temperature  $T_2 = -200 \, ^{\circ}C$ 

Longitudinal cross-section of the material  $A = 4 cm^2$ 

Load F = 480 N

### **Assumptions**

The material is homogeneous and isotropic

#### What is asked?

- a) Total longitudinal elongation
- b) Stress energy density of the bar
- c) Stress induced in the bar by cooling

#### Principles and formula

Hooke's law

$$\sigma = E\varepsilon \tag{1}$$

Where E is the Young modulus of the material that composes the bar,  $\sigma$  is the normal stress resulting from the load applied to the bar, and  $\varepsilon$  is the normal strain resulting from the load applied to the bar.

Strain definition

$$\varepsilon = \frac{\Delta L}{L_0} \tag{2}$$

 $\Delta L$  is the normal deformation, and  $L_0$  is the initial length of the bar.

Stress definition

$$\sigma = \frac{N}{A} \tag{3}$$

*N* is the internal force, and *A* is the cross section area.

Thermal strain

$$\varepsilon_{Th} = \alpha \Delta T \tag{4}$$

 $\varepsilon_{Th}$  is the thermal strain,  $\alpha$  is the thermal expansion coefficient of the material,  $\Delta T$  is the temperature variation between two states in thermal equilibria.

Strain energy density of the bar

$$U_0 = \frac{1}{2}E(\varepsilon_{tot} - \varepsilon_{Th})^2 \tag{5}$$

 $U_0$  is the strain energy density of the bar,  $\varepsilon_{tot}$  is the total strain of the bar, and  $\varepsilon_{Th}$  is the thermal strain of the bar.



#### Calculations

a) In this section  $\Delta T = T_1 - T_0$ . We calculate the load deformation thanks to Hooke's law and stress and deformation definitions.

$$\sigma = E \cdot \varepsilon \to \frac{N}{A} = \frac{F}{A} = \frac{\Delta L}{L_0} \cdot E \to \Delta L = \frac{F \cdot L_0}{E \cdot A} \tag{6}$$

Using Eq. (6) and Eq. (4), we can calculate the total deformation, which is caused by both mechanical and thermal load. We apply the superposition principle and:

$$\Delta L_{tot} = \Delta L + \varepsilon_{Th} L_0 = \frac{F \cdot L_0}{E \cdot A} + \alpha L_0 \Delta T \tag{7}$$

b) We can now rewrite Eq. (5) in terms of strain:

$$\varepsilon_{tot} = \frac{\Delta L_{tot}}{L_0} = \frac{\Delta L}{L_0} + \varepsilon_{Th} = \frac{F}{E \cdot A} + \alpha \Delta T = \varepsilon_F + \varepsilon_{Th}$$
 (8)

where  $\varepsilon_{Th}$  is the strain caused by the thermal expansion and  $\varepsilon_F$  is the strain caused by the load F. Therefore the strain energy is given by:

$$U_0 = \frac{1}{2}E(\varepsilon_{tot} - \varepsilon_{Th})^2 = \frac{1}{2}E\varepsilon_F^2 = \frac{1}{2}E\left(\frac{F}{E \cdot A}\right)^2$$
(9)

c) In this section  $\Delta T = T_2 - T_1$ . Since the beam is clamped, we can see that the  $\varepsilon_{tot} = 0$ , thus:

$$\varepsilon_{tot} = \varepsilon_{mech} + \varepsilon_{Th} = \frac{\sigma}{E} + \varepsilon_{Th} = 0 \rightarrow \sigma = -E\varepsilon_{Th} = -E\alpha\Delta T$$
 (10)

State your answer

a)

$$\Delta L_{tot} = \frac{F \cdot L_0}{E \cdot A} + \alpha (\Delta T) L_0 = \frac{480 \cdot 10 \cdot 10^{-2}}{40 \cdot 10^9 \cdot 4 \cdot 10^{-4}} + 10 \cdot 10^{-6} \cdot 475 \cdot 10 \cdot 10^{-2} = 478 \,\mu m$$
(11)

b)

$$U_0 = \frac{1}{2} 40 \cdot 10^9 \cdot \left(\frac{480}{40 \cdot 10^9 \cdot 4 \cdot 10^{-4}}\right)^2 = 18 \frac{J}{m^3}$$
 (12)

c)

$$\sigma = -40 \cdot 10^9 \cdot 10 \cdot 10^{-6} \cdot (-700) = 280 \, MPa \tag{13}$$

The mechanical stress that builds up in the bar when cooling down is 280 MPa tensile.



## Exercise 3b.2 - Plane Stress

A square plate of width b and thickness t is loaded by normal forces  $F_x$  and  $F_y$ , and by shear forces V, as shown in Figure 3b.2. These forces produce uniformly distributed stresses acting on the side faces of the plate. Calculate the change in the volume  $\Delta V = V_{final} - V_{initial}$  of the plate and strain energy U stored in the plate if the dimensions are  $b = 600 \, mm$  and  $t = 40 \, mm$ , the plate is made of magnesium with  $E = 45 \, GPa$  and v = 0.35, and the forces are  $F_x = 480 \, kN$ ,  $F_y = 180 \, kN$ , and  $V = 120 \, kN$ .

Formula for strain energy density in two dimensions.

$$u_0 = \frac{1}{2} \left( \sigma_x \varepsilon_x + \sigma_y \varepsilon_y + \tau_{xy} \gamma_{xy} \right) \tag{1}$$

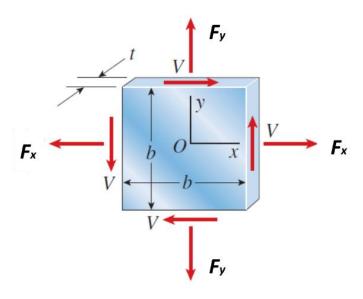


Figure 3b.2 | Loads on a cube



#### Given:

A block with multiple normal and shear loads, see in figure 3b.2:

 $b = 600 \, mm$ 

t = 40 mm

E = 45 GPa

 $\nu = 0.35$ 

 $F_x = 480 \ kN$ 

 $F_{\rm v} = 180 \ kN$ 

 $V = 120 \, kN$ 

#### What is asked:

- a) Change in volume,  $V_{final} V_{initial}$ .
- b) Strain energy stored in the block.

#### **Solution:**

a) The stresses in the material can be directly calculated from the loads, i.e.:

$$\sigma_x = \frac{N_x}{A} = \frac{F_x}{A} = \frac{F_x}{bt}; \sigma_y = \frac{F_y}{A} = \frac{F_y}{bt}; \tau_{xy} = \frac{V}{A} = \frac{V}{bt}$$
 (2)

As we know from theory, applying the strain equations from the compliance matrix to Eq. (2) we can derive the following equation:

$$\frac{V_{final} - V_{initial}}{V_{initial}} = \frac{(1 - 2\nu)}{E} \left(\sigma_x + \sigma_y + \sigma_z\right) \tag{3}$$

Inserting values for the initial volume  $V_{initial} = b^2 \cdot t$ ,  $\sigma_x$ , and  $\sigma_y$  into Eq. (3) we obtain the following:

$$V_{final} - V_{initial} = \left(b^2 \cdot t\right) \cdot \left[\frac{(1 - 2\nu)}{E \cdot (t \cdot b)}\right] \left(F_x + F_y\right) = b \frac{(1 - 2\nu)}{E} \left(F_x + F_y\right) \tag{4}$$

$$V_{final} - V_{initial} = 600 \ mm \cdot \frac{1 - 2 \cdot 0.35}{45 \ GPa} (480 \ kN + 180 \ kN) = 4 \cdot 660 \cdot 10^{-9} \ m^3$$
 (5)

$$V_{final} - V_{initial} = 2.64 \cdot 10^{-6} m^3 \tag{6}$$

b) Using Eq. (1) and substituting strains with stresses with 3D Hooke's law, we are left with the following:

$$u_0 = \frac{1}{2E} \left( \sigma_x^2 + \sigma_y^2 - 2\nu \sigma_x \sigma_y \right) + \frac{\tau_{xy}^2}{2G} \tag{7}$$

$$G = \frac{E}{2(1+\nu)} \tag{8}$$

Inserting the stresses from Eq. (2) into (7), we get the following:

$$u_0 = \frac{1}{2E} \left( \left( \frac{F_x}{t \cdot b} \right)^2 + \left( \frac{F_y}{t \cdot b} \right)^2 - 2\nu \left( \frac{F_x}{t \cdot b} \right) \left( \frac{F_y}{t \cdot b} \right) \right) + \frac{1}{2G} \left( \frac{V}{t \cdot b} \right)^2$$
 (9)



$$u_0 = 4653 Pa = 4653 J/m^3 (10)$$

Remember that  $u_0$  is the strain energy *density*, so the final result is:

$$U = u_0 \cdot V_{initial} = 4653 \frac{J}{m^3} \cdot 600^2 \ mm^2 \cdot 40 \ mm = 67 \ J \tag{11}$$

Also keep in mind that the relative change in volume is very small so it is not going to make a difference whether we consider  $V_{initial}$  or  $V_{final}$  in Eq. (11).



## Exercise 3b.3 - Hybrid stiffness - 3D structure

A block of rubber (R on Figure 3b.3) is confined in a slot inside a steel block (S on Figure 3b.3). A uniform pressure  $p_0$  applied on the top of the rubber block induces a deformation. The rubber's Young's modulus E and the rubber's Poisson's ratio  $\nu$  are known.

- a) Give an expression for the pressure along x axis on the block induced by  $p_0$  and calculate its value
  - NB We will neglect friction effects
- b) Give an expression for the dilatation e of the rubber and calculate its value
  - NB The dilatation is also called the relative volume variation, i.e.  $e = \frac{\Delta V}{V}$
- c) Find the strain-energy density  $u_0$  of the rubber

Numerical values:  $p_0 = 5.0$  MPa; E = 15.0 GPa; v = 0.50

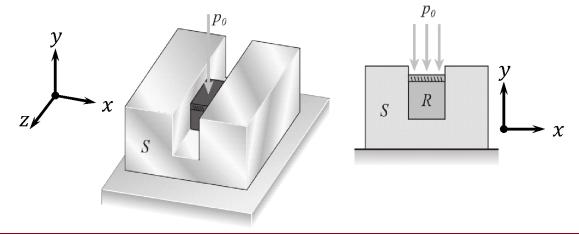


Figure 3b.3 | Block of rubber in a steel block



## What is given?

Pressure  $p_0 = 5.0 MPa$ 

Young's modulus E = 15.0 GPa

Poisson's ratio  $\nu = 0.50$ 

#### What is asked

- a) A formula for the lateral pressure on the block induced by  $p_0$  and calculate its value
- b) A formula for the dilatation e of the rubber and calculate its value
- c) The strain-energy density  $u_0$  of the rubber

#### **Equations** required

We will use the generalized Hooke's law, or compliance matrix:

$$\varepsilon_{x} = \frac{1}{E} \left( \sigma_{x} - \nu \left( \sigma_{y} + \sigma_{z} \right) \right); \ \gamma_{xy} = \frac{\tau_{xy}}{G}; etc \dots$$
 (1)

Where E is the Young's modulus of the material, v is the Poisson's ratio,  $\varepsilon_x$  is the axial strain in the x-direction, and  $\sigma_x$  is the normal stress parallel to the x-axis,  $\tau_{xy}$  and  $\gamma_{xy}$  are the shear stress and strain on the plane xy, and G is the shear modulus of the material. We then define the strain energy density in three dimensions:

$$u_0 = \frac{1}{2E} \left( \sigma_x^2 + \sigma_y^2 + \sigma_z^2 \right) - \frac{\nu}{E} \left( \sigma_x \sigma_y + \sigma_x \sigma_z + \sigma_y \sigma_z \right) + \frac{1}{2G} \left( \tau_{xy}^2 + \tau_{yz}^2 + \tau_{zx}^2 \right)$$
 (2)

We can also write the volume relative variation in relation with the stress components:

$$\frac{\Delta V}{V} = \frac{1 - 2\nu}{E} (\sigma_x + \sigma_y + \sigma_z) \tag{3}$$

#### Calculations

a) The pressure is opposed to the internal stress of the material. We are looking for the pressure *p* along the *x*-direction.

$$p_x = -\sigma_x \tag{4}$$

We have been given the pressure in the y-direction.

$$p_0 = -\sigma_v \tag{5}$$

Then, no stress is induced in the z-direction, and being clamped, no strain can occur in the x direction:

$$\sigma_z = 0; \varepsilon_x = 0 \tag{6}$$

We apply the general Hooke's law.

$$\varepsilon_{x} = \frac{1}{E} \left( \sigma_{x} - \nu (\sigma_{y} + \sigma_{z}) \right) \tag{7}$$

i.e. using (4), (5) and (6) in (7):

$$\varepsilon_{x} = 0 = -p_{x} - \nu(-p_{0}) \tag{8}$$

Thus,

$$p_x = \nu p_0 \tag{9}$$



b)

$$\frac{\Delta V}{V} = \frac{1 - 2\nu}{E} (\sigma_x + \sigma_y + \sigma_z) \tag{10}$$

i.e.

$$\frac{\Delta V}{V} = \frac{1 - 2\nu}{E} (-p_x - p_0) = -\frac{(1 - 2\nu)(1 + \nu)p_0}{E}$$
 (11)

c) Using the formula of the strain energy, we substitute the different components of stress in all the directions. There is no shear due to the wall constraint and only the *y* direction contributes:

$$u_0 = \frac{1}{2E}(1 - v^2)p_0^2 \tag{12}$$

#### State your answer

a) The lateral pressure is:

$$p_x = \nu p_0 = 2.5 \, MPa \tag{13}$$

b) The relative change in volume is:

$$\frac{\Delta V}{V} = -\frac{(1 - 2\nu)(1 + \nu)p_0}{E} = 0 \tag{14}$$

There is no volume variation. The rubber keeps the same properties.

c) The strain energy density (in 3D) is:

$$u_0 = \frac{1}{2E}(1 - v^2)p_0^2 = \frac{1}{2}\frac{1 - 0.5^2}{15 \cdot 10^9}5^2 \cdot 10^{12} Pa = 625 Pa = 625 \frac{N}{m^2} = 625 \frac{J}{m^3}$$
 (15)



## Exercise 3b.4 - Bars and spring in series

A system 1) is composed of two different bars while a similar system 2) is instead formed by a spring and a bar, as shown in the Figure 3b.4 Both systems are loaded with forces in A, B and C, and the materials are considered isotropic.

### a) Draw the Free Body Diagram of the two systems

Provided the numerical values for the systems:

1) 
$$A = 3 cm^2$$
,  $A_* = 2 cm^2$ ,  $E = 25 GPa$ ,  $L = 10 cm$ ,  $F_1 = 30 kN$ ,  $F_2 = 45 kN$ ,  $F_3 = 75 kN$ 

2) 
$$A = 3 cm^2$$
,  $E = 25 GPa$ ,  $L = 10 cm$ ,  $F_1 = 30 kN$ ,  $F_2 = 45 kN$ ,  $F_3 = 75kN$ ,  $F_3 = 1.10^8 \frac{kg}{c^2}$ 

### b) Calculate the deformation of the two different systems

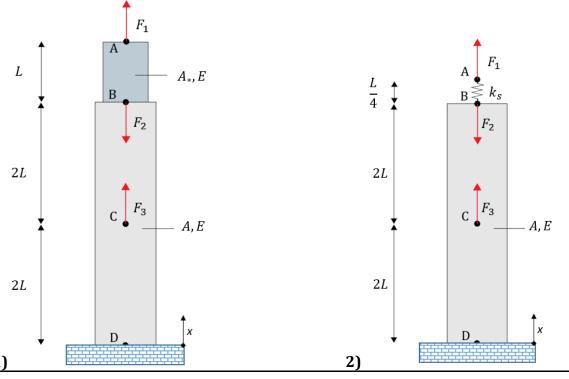
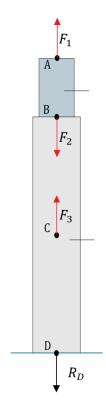


Figure 3b.4 | Composite posts: 1) bar/bar and 2) spring/bar



a) The FBD is equal for both the systems since the spring can be seen as a bar



The structure can be divided as shown in figure 3b4.1 in order to calculate the internal forces:

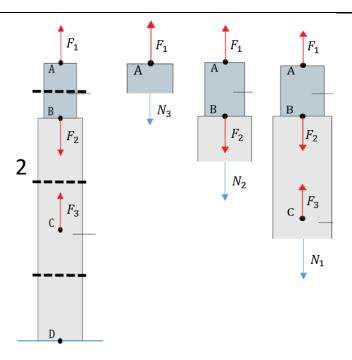


Figure 3b.4.1 | Composite posts: cuts to be done

What are the Eqs. that are required?

The stiffness of a segment CD:



$$k_{CD} = \frac{AE}{L_{CD}} \tag{1}$$

The stiffness of a segment CB:

$$k_{CB} = \frac{AE}{L_{CB}} \tag{2}$$

The stiffness of a segment AB, system a):

$$k_{AB} = \frac{A_* E}{L_{AB}} \tag{3}$$

While for system 2) is it equal to the stiffness of the spring  $k_s$ 

Where A and  $A_*$  are the cross-section area of segments,  $L_{AB}$ ,  $L_{CB}$  and  $L_{CD}$ , are the length and E the Young's modulus.

The internal force of a segment with respect to the displacement:

$$N = k\Delta \tag{4}$$

#### Find Reaction at Point D

$$\sum F_x = 0 \tag{5}$$

$$-R_D + 75kN - 45kN + 30kN = 0 \rightarrow R_D = 60kN$$
 (6)

b) The deformation of the two systems can be calculated using

$$\delta_a = \sum_{i} \frac{N_i L_i}{A_i E} = \frac{1}{E} \left( \frac{N_1 * 2L}{A} + \frac{N_2 * 2L}{A} + \frac{N_3 * L}{A_*} \right)$$
 (7)

$$\delta_b = \frac{1}{E} \left( \frac{N_1 * 2L}{A} + \frac{N_2 * 2L}{A} \right) + \frac{N_3}{k_s} \tag{8}$$

From figure 3b.4.1 can be seen that:

$$N_1 = 60 kN$$

$$N_2 = -15 kN$$

$$N_3 = 30 kN$$

Plugging the numbers in (7) and (8) can be calculated the displacement:

$$\delta_a = \frac{1}{25 * 10^9} \left( \frac{60 * 10^3 * 0.2}{3 * 10^{-4}} - \frac{15 * 10^3 * 0.2}{3 * 10^{-4}} + \frac{30 * 10^3 * 0.1}{2 * 10^{-4}} \right) = 0.0018 = 1.8 \, mm \tag{9}$$

$$\delta_b = \frac{1}{25 * 10^9} \left( \frac{60 * 10^3 * 0.2}{3 * 10^{-4}} - \frac{15 * 10^3 * 0.2}{3 * 10^{-4}} \right) + \frac{30 * 10^3}{10^8} = 0.0015 = 1.5 \ mm \tag{10}$$



## Exercise 3b.5 - Composed post

A post is composed of two different elements: a cube of height 3L between C and E (Young's modulus  $E_{CE}$ ) and a square based tempered post with a of height 6L between A and C (Young's modulus  $E_{AC}$ ). As shown in figure 3b.5, the section varies from A (side length 2L) to C (side length 3L) and two forces are applied to the system at point A and C. The amplitude of the force at point C is 2F and the amplitude of the force at point A is F. The materials are considered isotropic.

- a) Draw the Free Body Diagram of the system and calculate the reaction force(s)
- b) Calculate the value of the stress and strain of the post at section D
- c) Calculate the value of the stress and strain of the post at section B
- d) Calculate the deformation of the segment AC

$$Mathematical\ Hint: \int \frac{dx}{(a+bx)^2} = -\frac{1}{b(a+bx)}$$

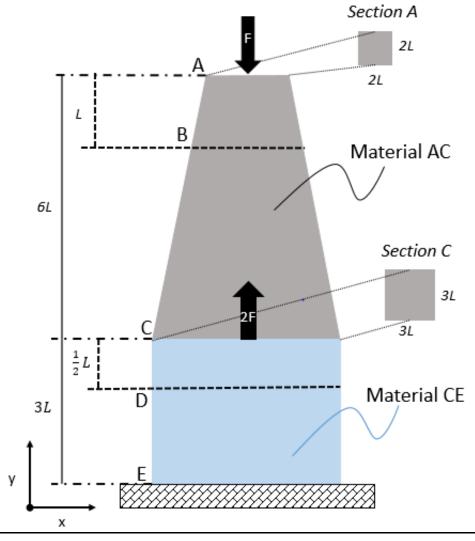
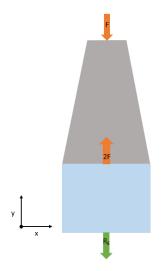


Figure 3b.5 | Composed post



### a) Draw the Free Body Diagram of the system and calculate the reaction force(s).

Apply force equilibrium Eq. to the entire structure and evaluate R<sub>E</sub>.



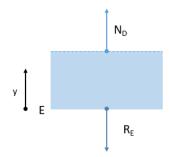
$$\sum F_{y} = 0 \to -R_{E} + 2F - F = 0 \to R_{E} = F$$
 (1)

#### b) Calculate the value of the stress and strain of the post at section D

For the segment from E to D the area of the section is:

$$A = (3L)^2 = 9L^2$$

#### **Segment DE**



$$\sum F_{y} = 0 \to -R_{E} + N_{D} = 0 \quad N_{D} = R_{E} = F$$
 (2)

For the evaluation of the stress and the strain of the post at section D

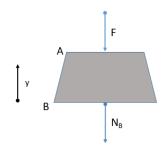
$$\sigma_D = \frac{N_D}{A_D} = \frac{F}{9L^2} \tag{3}$$

Strain is:

$$\varepsilon_D = \frac{\sigma_D}{E_{CE}} = \frac{F}{9L^2 E_{CE}} \tag{4}$$



## c) Calculate the value of the stress and strain of the post at section B Segment AB



$$\sum F_{y} = 0 \to -F + -N_{B} = 0 \to N_{B} = -F$$
 (5)

For the segment from A to C the dimension of the square follows these formulas:

$$l(y) = L\left(2 + \frac{1}{6}\frac{y}{L}\right) \tag{6}$$

$$A(y) = L^2 \left( 2 + \frac{1}{6} \frac{y}{L} \right)^2 \tag{7}$$

or

$$l(y) = L\left(3 - \frac{1}{6}\frac{y}{L}\right)$$
$$A(y) = L^2\left(3 - \frac{1}{6}\frac{y}{L}\right)^2$$

The area of section B is:

$$A(L) = L^2 \left( 2 + \frac{L}{6L} \right)^2 = \frac{169}{36} L^2 \tag{8}$$

or

$$A(5L) = L^2 \left( 3 - \frac{5L}{6L} \right)^2 = \frac{169}{36} L^2$$

For the evaluation of the stress and the strain of the post at section B

$$\sigma_B = \frac{N_B}{A_B} = \frac{-F}{\frac{169}{36}L^2} = -\frac{36F}{169L^2} \tag{9}$$

Strain is:

$$\varepsilon_B = \frac{\sigma_B}{E_S} = -\frac{36F}{169L^2E_{AC}} \tag{10}$$



### d) Calculate the deformation of the segment AC

Since the section varies along the axis is necessary to integrate between the tip of the post A and the section C.

The elongation can be evaluated with:

$$d\delta = \frac{-Fdy}{E_{AC}A(y)} \tag{11}$$

By integrating (from A to C):

$$\delta_{AC} = \int_0^{6L} \frac{-Fdy}{E_{AC}A(y)} \tag{12}$$

$$\delta_{AC} = \frac{-F}{E_{AC}L^2} \int_0^{6L} \frac{dy}{\left(2 + \frac{y}{6L}\right)^2} = -\frac{F}{E_{AC}L^2} \left[ -\frac{6L}{2 + \frac{y}{6L}} \right]_0^{6L}$$

$$= -\frac{F}{E_{AC}L^2} (-2L + 3L) = -\frac{FL}{E_{AC}L^2} = -\frac{F}{E_{AC}L}$$
(13)

Or (from C to A)

$$\delta_{AC} = \int_0^{6L} \frac{-F dy}{E_{AC} A(y)}$$

$$\delta_{AC} = -\frac{F}{E_{AC} L^2} \int_0^{6L} \frac{dy}{\left(3 - \frac{y}{6L}\right)^2} = -\frac{F}{E_{AC} L^2} \left[ -\frac{6L}{3 - \frac{y}{6L}} \right]_0^{6L} = -\frac{F}{E_{AC} L^2} (-2L + 3L) = -\frac{FL}{E_{AC} L^2} = -\frac{F}{E_{AC} L}$$

The segment AC is compressed and shortens of a quantity equal to:  $\frac{F}{E_{AC}L}$